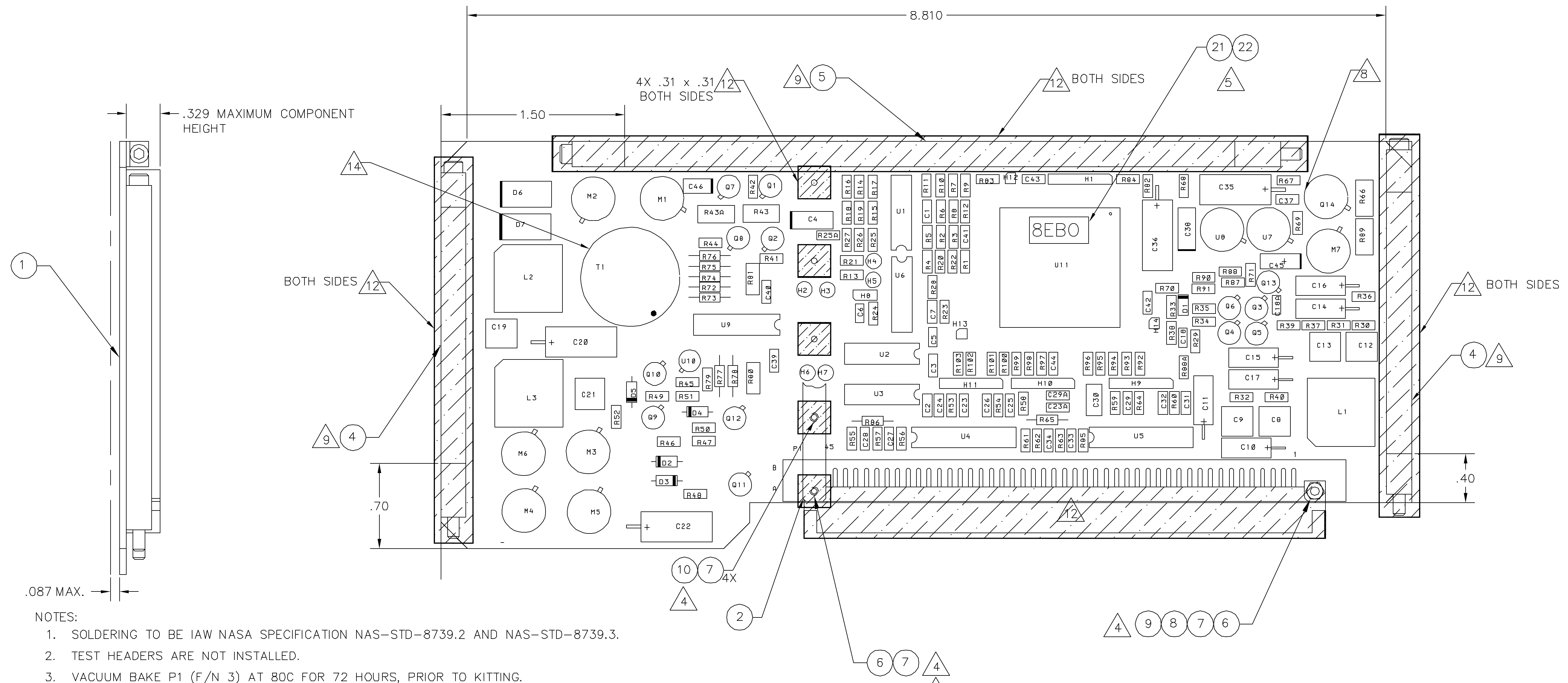


ECO		REVISIONS		
REV	DESCRIPTION	CHECKED	APPROVED	DATE
30-012	A	EAB	R. FOSTER	9/3/02
30-041	B	EAB	R. FOSTER	3/21/03
30-071	C	EAB	R. FOSTER	11/13/03
30-0	D			



NOTES:

- SOLDERING TO BE IAW NASA SPECIFICATION NAS-STD-8739.2 AND NAS-STD-8739.3.
- TEST HEADERS ARE NOT INSTALLED.
- VACUUM BAKE P1 (F/N 3) AT 80C FOR 72 HOURS, PRIOR TO KITTING.
- REMOVE EXISTING HARDWARE AFTER CONFORMAL COAT. INSTALL F/N 6,7,8,9 AND 2 AS SHOWN. TORQUE #2-56UNC SCREW TO 32 IN-OZ. SPOTBOND WITH F/N 17. MIX RATIO: 4g 828V125/2g ALUMINA POWDER.
- PLACE CHECKSUM LABEL ON U11 USING F/N 21 AND COAT WITH F/N 22.
- SPOTBOND CORNERS OF U11 AND C10,11,14-17,20,22,35,36, D6, D7, L1-L3 AND T1 USING F/N 16 PRIOR TO CONFORMAL COAT.
- PARTS TO RECEIVE CURSORY VISUAL INSPECTION PRIOR TO KITTING AND INSTALLATION.
- VERIFY C37 IS NOT SHORTED TO VIA UNDERNEATH CAPACITOR.
- EPOXY F/N 4, 5 AND 20 TO F/N 1 USING F/N 13, AFTER CONFORMAL COAT. MOUNTING DIMENSIONS SPECIFIES STARTING LOCATION FOR EPOXYING FIXED PORTION OF WEDGELOK ASSY.
- ITEM IS STATIC SENSITIVE. HANDLE IAW MIT PROCEDURE 99-01003 OR EQUIVALENT.
- MEASURE UNMARKED CAPACITORS PRIOR TO KITTING. BAG AND TAG.
- MASK FOR CONFORMAL COAT AS INDICATED. PRIOR TO SHIPMENT FOR CONFORMAL COAT; CLEAN, INSPECT AND AIR BAKE FOR 1 HOUR AT 65° C. BAG AND SEAL.
- CONFORMAL COAT ASSEMBLY WITH F/N 14.
 - CURE AT 25C, AMBIENT PRESSURE FOR 8 HOURS.
 - REMOVE MASKING AND MASKING RESIDUE PRIOR TO VACUUM BAKE.
 - VACUUM BAKE AT 65C, 10⁻⁵ TORR FOR 24 HOURS.
- TORQUE F/N 19 TO 80 IN OZ AND SPOTBOND WITH F/N 17.

UNLESS OTHERWISE SPECIFIED DIMENSIONS ARE IN INCHES. TOLERANCE: ANGLES +/- 1° 3 PLACE DECIMALS +/- .005 2 PLACE DECIMALS +/- .01		NAME		DATE		Massachusetts Institute of Technology Center for Space Research ASTRO-E2	
MATERIAL N/A		DRAWN	M. SMITH	1/29/02		CCA, THERMAL CONTROL	
FINISH N/A		CHECKED	EAB	8/29/02			
30-20000		ASTRO-E2	APPROVED	R. FOSTER	9/3/02		SIZE D 80230
NEXT ASSEMBLY		USED ON	RELEASED	D. GAGE	9/3/02		
APPLICATION		CAD FILE		SCALE 1:1		DWG NO. 30-20200	REV D